

Valve Operated Low-Leakage Cryogenic-Connector T513 | Joshua Leary | Juan Valencia | Jackson Herrod | Mika Kuschnitzky Advisor: Mark Vanderlaan | Sponsors: Rachel McCauley, Shawn Brechbill

Key Goals

- **Refuel Rocket Ship**
- Versatility in Application
- **Operable without Machinery**
- Low Loss of Fuel
- **Ergonomic for Astronaut**

Testing

Test 1: Valve Activation Force Check

Test 2: Gaseous Helium Leak Check

Test 3: Cryogenic Nitrogen Flow

Test 4: Assembly Cycle Testing

V.O.L.C

Objective

Design, build, and test a cryogenic connector interface with a focus on the seal/joint design for refueling to support future missions on the moon.





Marshall Space Flight Center

Targets

- Connector Diameter: 1.00 1.20 in
- Volumetric Flow Rate: $0.100 \frac{in^3}{...}$ min
 - Leakage: 1% of flow
 - Activation Force: < 48 N

Future Development

- Mitigate Solar Radiation
- Mitigate Regolith Contamination
- **Design Efficient Shipping Container**
- **Design Mass Production Assembly**