

# Team 508 SAE Aero Design: Geometric Integration

**EML 4552C** 

# **Team Members**





Jacob Pifer
Project Manager
Manufacturing Engineer



Lauren Chin
Lift & Control Surfaces Engineer
Meeting Coordinator
CAD Engineer



Joseph Figari
Fuselage and Payload Engineer
Financial Coordinator
CAD Engineer

# **Sponsors**





Florida Space Grand Consortium Financial Sponsor



Seminole RC Club
Equipment Provider



Shayne McConomy, PhD
Faculty Sponsor



# Advisors





Simone Hruda, PhD Faculty Advisor



**Eric Adams**Fablab Supervisor

# **Project Objective**

- The objective of this project is to design and manufacture a 3D printed remote control airplane within the rules of the SAE Aero Design Competition
- It will be able to take off, complete the needed flight path, and land while carrying the required cargo



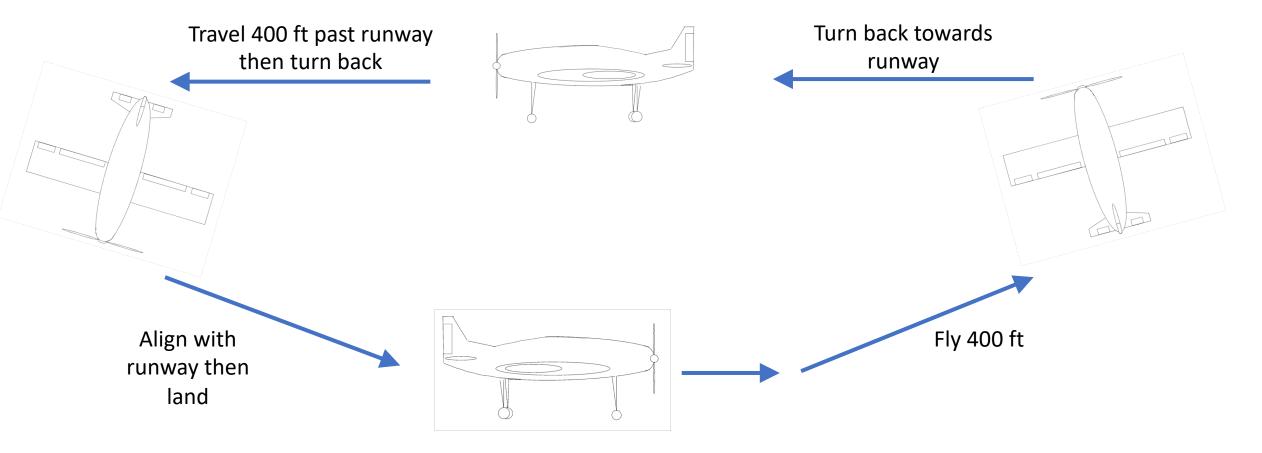


# **Project Brief Summary**

- > The plane will not be flown at the competition in March
- > Team is still a part of the competition
- > Plane will still be built within competition rules
- > Test flight will be done in Tallahassee
- ➤ If necessary and time permits, design changes will be made and a second flight will take place









## **Material Selection**

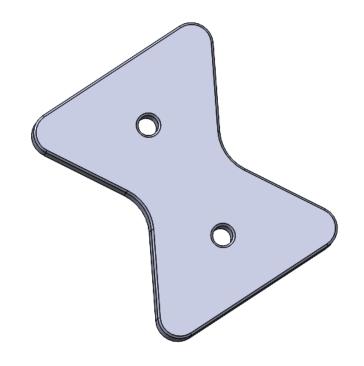
- ➤ House of Quality found weight to be most important design factor
- Two possible filaments could be used within budget and competition rules
  - > PLA
  - > LW-PLA
- ➤ Torsion and bending tests done to compare strengths
- Tests found PLA to be stronger but due to weight advantage LW-PLA was chosen





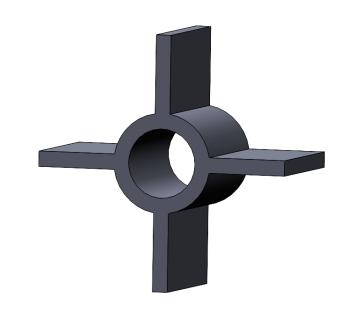
## **Assembling Methods: Bow Ties**

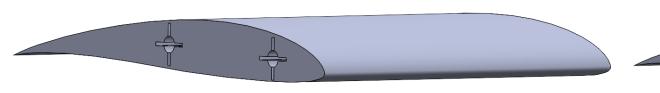
- ➤ Bowties will be used to connect parts of the fuselage
- > Typically three bowties per connection point
  - Only one needed to connect the tail sections
  - > One screw can be used to secure a bowtie



# Assembling Methods: Spars

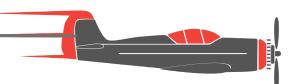
- > Plane uses spars to assemble the wings
- > Each wing set has two spars
- ➤ Small crosses are also used to help lock the wings together
- ➤ Ends of the spars are threaded so screws can be used to secure the parts

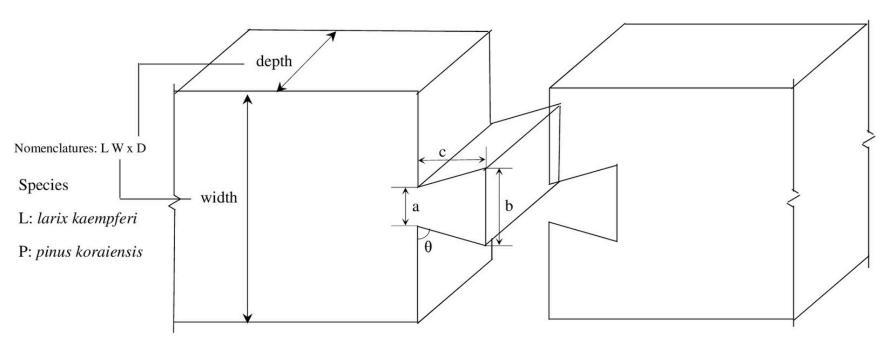






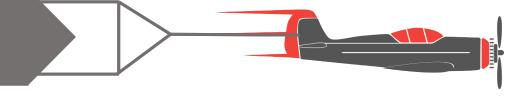
#### **Assembling Methods: Woodworking**

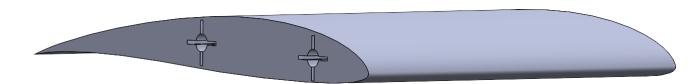




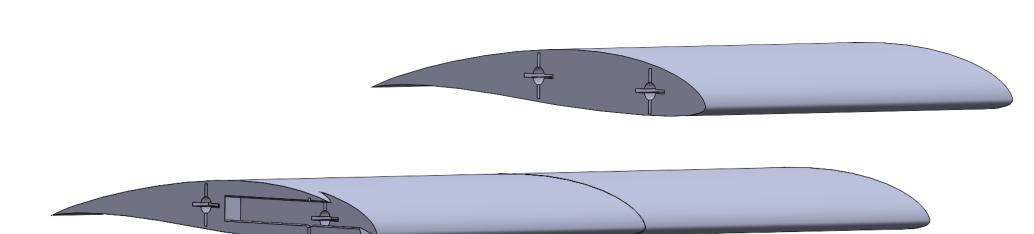
- ➤ The Dove Tail Connection
  - Consists of a male and female part
  - Prevents movement perpendicular to the connection
  - Prevents rotational movement

# **Canard Assembly**

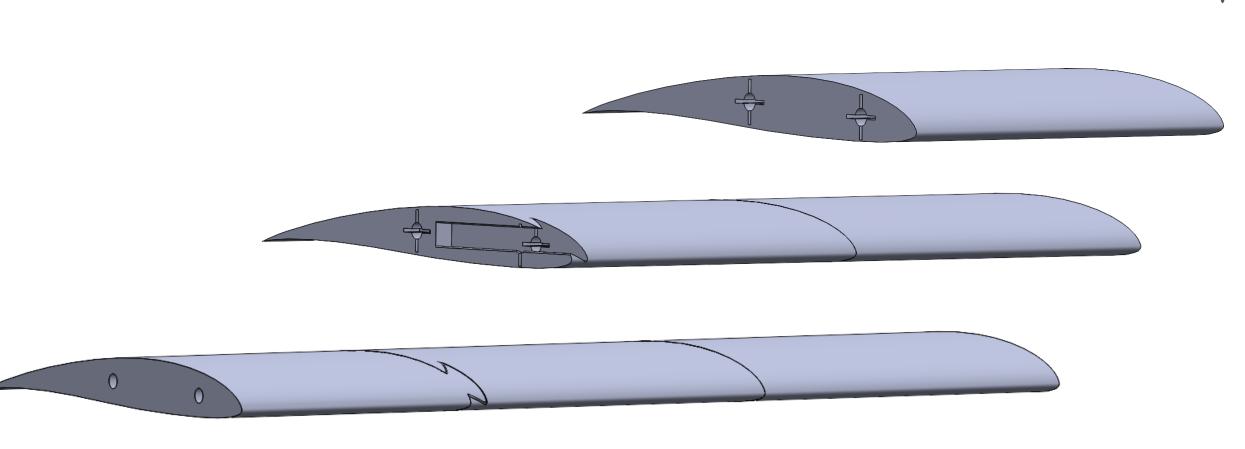


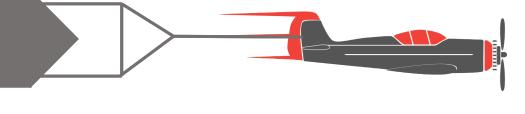


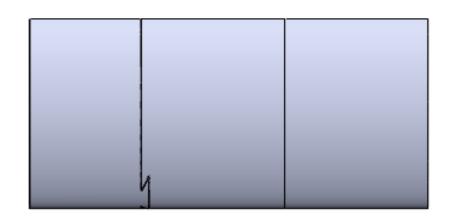
# **Canard Assembly**

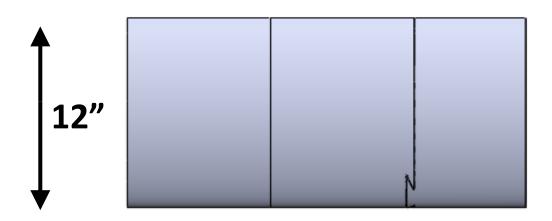


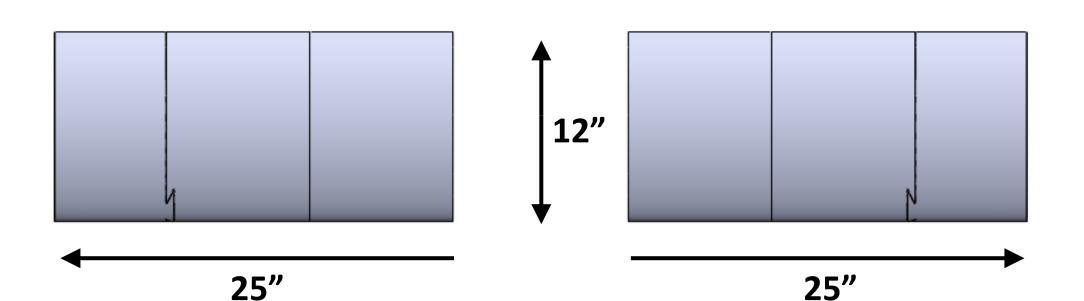
# Canard Assembly



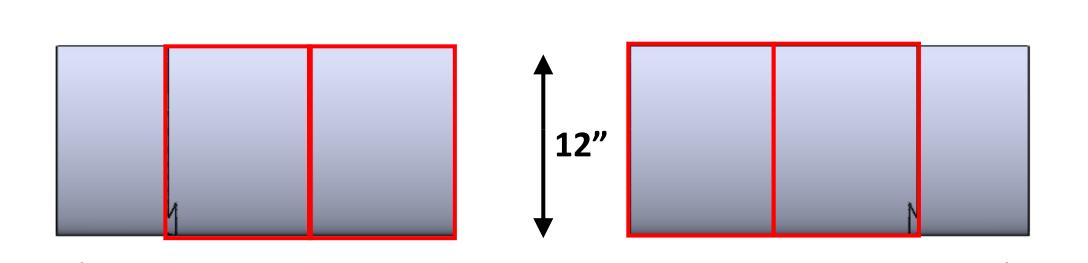








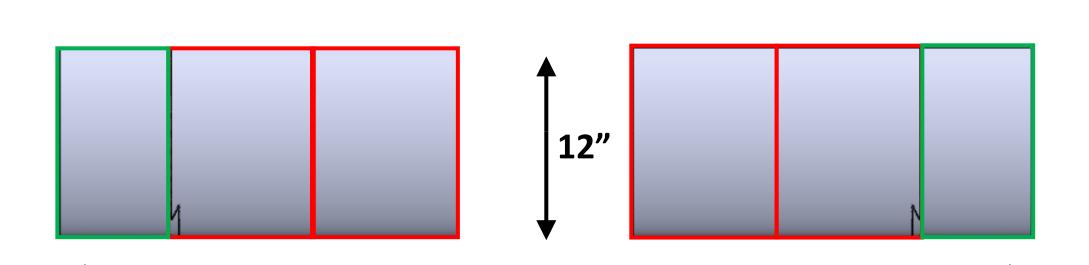
25"



Lauren Chin

**25**"

25"



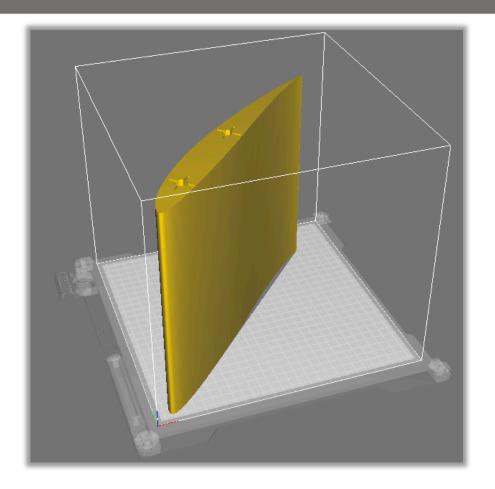
Lauren Chin

**25**"

#### **Printers**

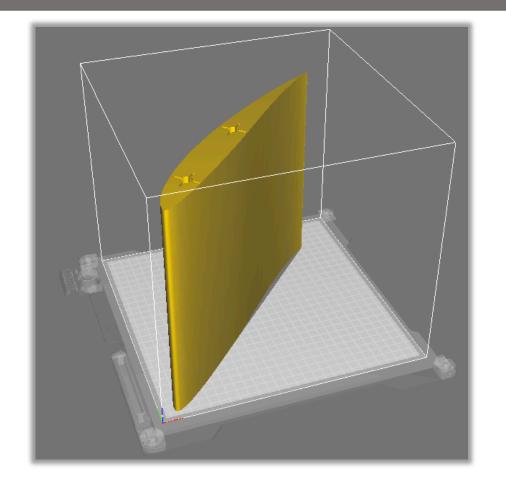
- ➤ Lulzbot TAZ printers are the main printers used
  - > Design Lab has two and the Innovation Hub has one
- > Any PLA parts will be printed at the Innovation Hub
  - > Small parts that DREMEL printers can make
- Cura-lulzbot used to queue prints
  - > Helps in estimating print time, material used, and needed printing orientations

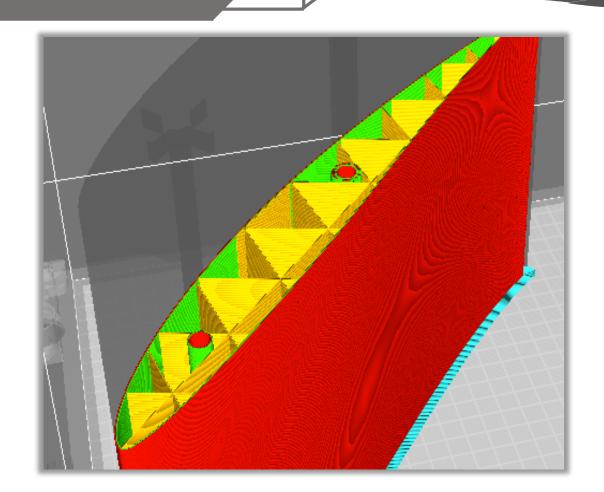
# **Printing Parts**





# **Printing Parts**





## In the Works



#### **Future Work**

> Design ways to secure the electronics inside the fuselage

- ➤ Build the landing gear
- ➤ Make remaining project orders
- ➤ Continue 3D printing the parts
- Serving la DC

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## **Key Takeaways**

- > Even though the teams aren't going to Lakeland, there will still be a test flight
- > The plane is still being built within competition rules
- > LW-PLA is here and printing has begun
- > All parts must be made with printing in mind

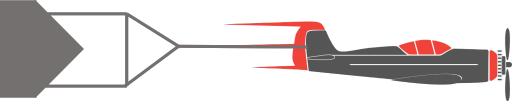
#### References



- ➤ 2021 Collegiate Series SAE Design Rulebook (2021). SAE Aero Design.

  <a href="https://www.saeaerodesign.com/cdsweb/gen/DocumentResources.aspx">https://www.saeaerodesign.com/cdsweb/gen/DocumentResources.aspx</a>
- ➤ Aguirre, N., Evans, L.,... Silver, Z. (2020). T513: SAE Aero Design Operations Manual. Team 513: SAE Aero Design East Competition, 47-56.

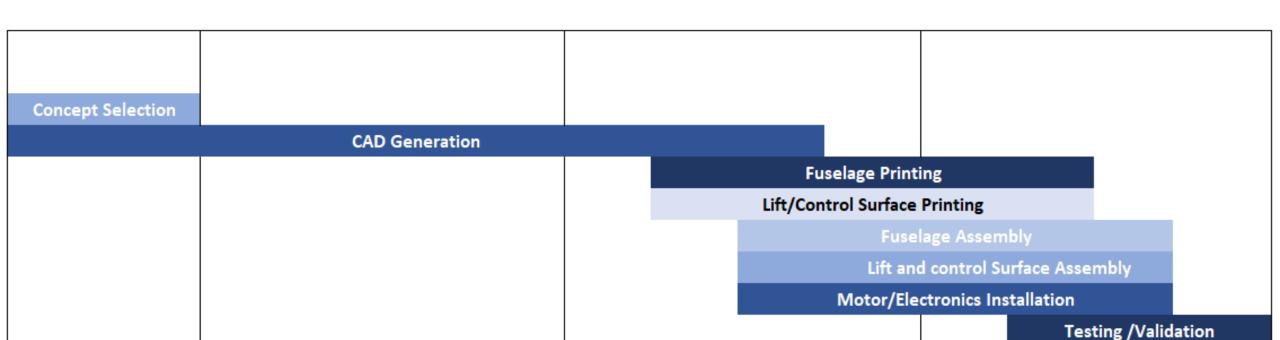
# **Fall Semester Timeline**



Preliminary Research												
				Concept Generation								
										Con	cept Select	ion
Aug 30- Sept 5	Sept 6-12	Sept 13 - 19	Sept 20-26	Sept 27- Oct 3	Oct 4-10	Oct 11- 17	Oct 18-24	Oct 25-31	Nov 1-7	Nov 8-14	Nov 18-21	Nov 22 -28

Sep-20 Oct-20 Nov-20

#### Winter & Spring Semester Timeline



Jan 10-16

Dec-20 Jan-21 Feb-21 Mar-21

Jan 17-23

Jan 24-30

Jan 31 - Feb6

Feb 7-13

Jacob Pifer

Feb 21-27 Feb 28- Mar 6

Feb 14-20

Dec 13-19

Dec 20-26

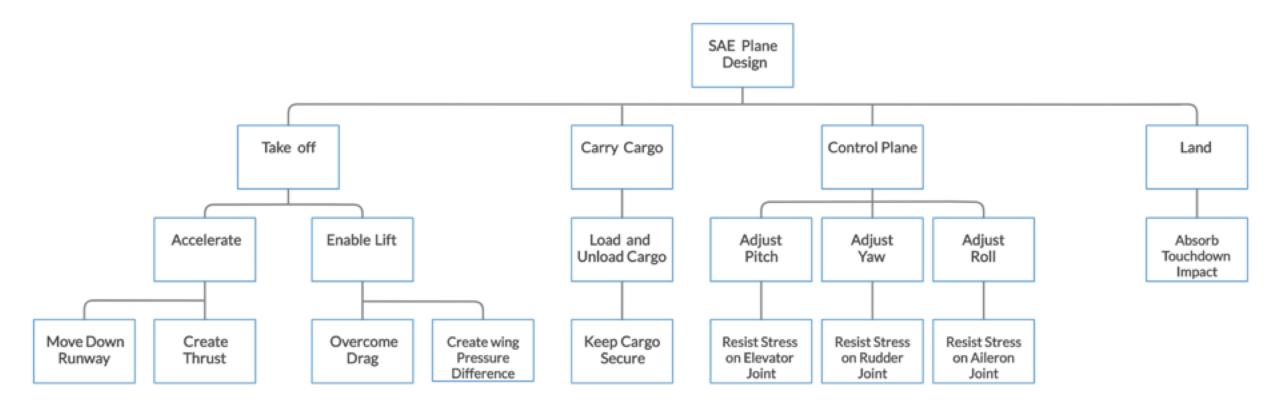
Dec 27-Jan2

Jan 3-9

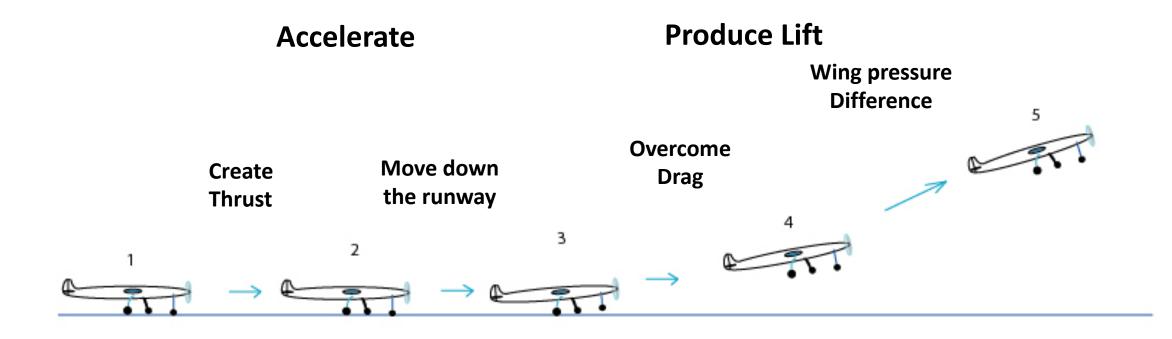
Dec 6-12

Nov 29- Dec 5

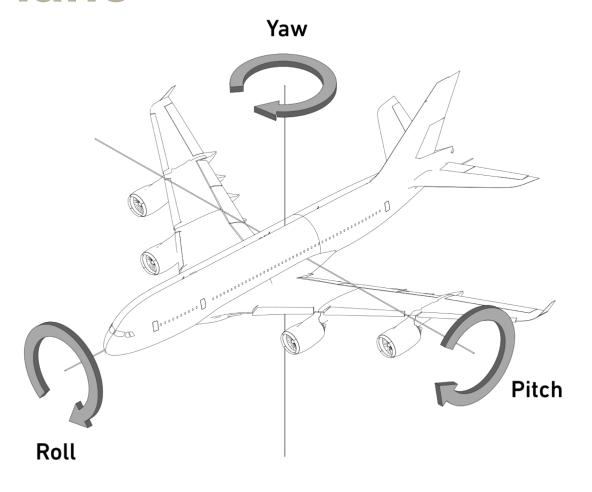
# **Functional Decomposition**



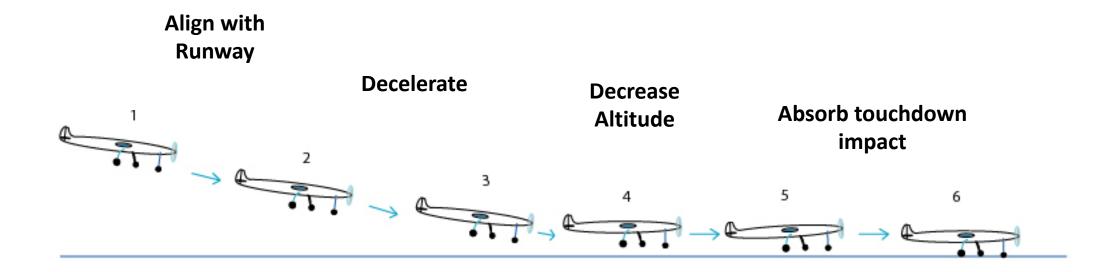
## **Takeoff**

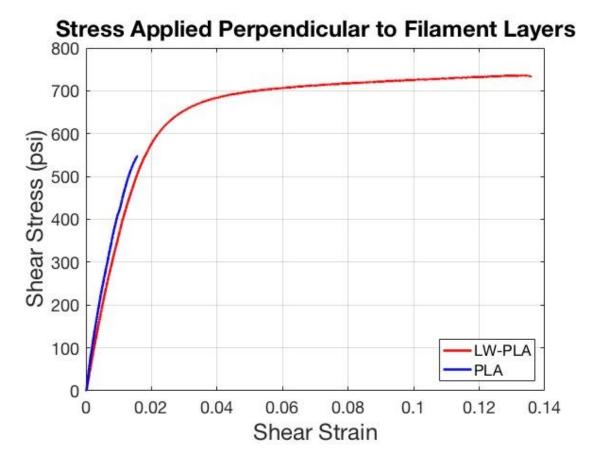


## **Control Plane**

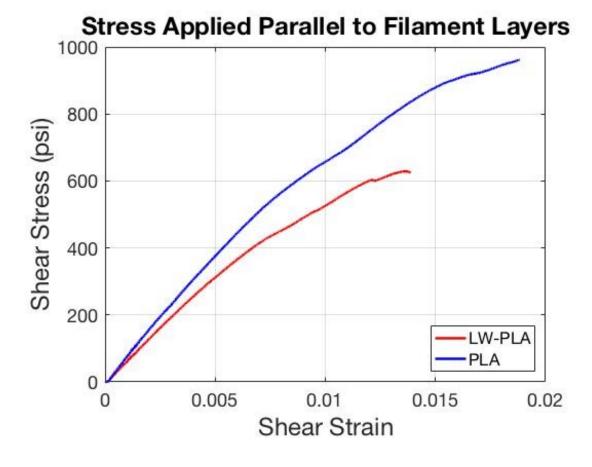


# Landing





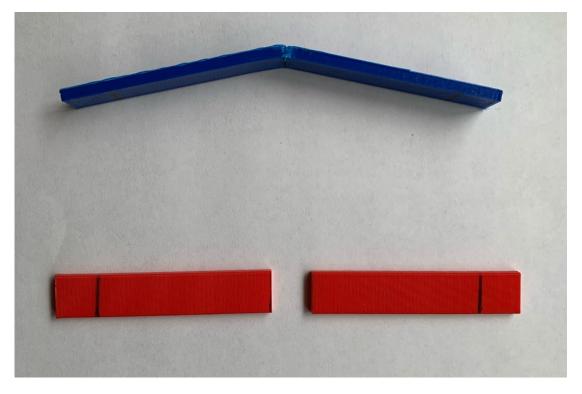
PLA failure stress: 412 psi LW-PLA failure stress: 552 psi



PLA failure stress: 721 psi

LW-PLA failure stress: 471 psi

#### **PLA**



Stress Perpendicular to layering direction: 3,360 psi Stress Parallel to layering direction: 8,350 psi

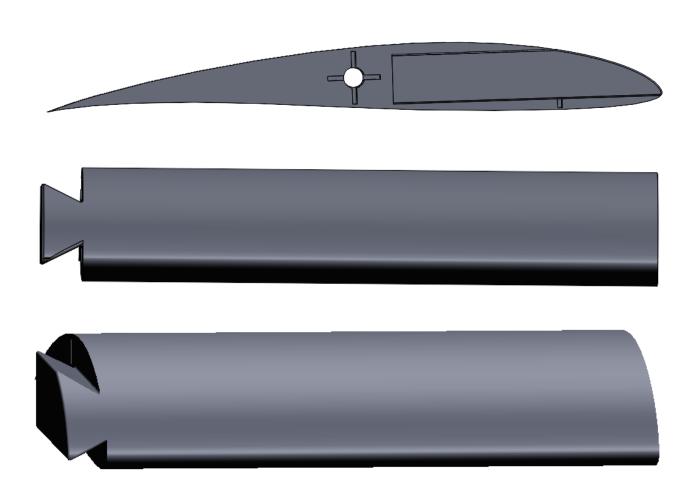
#### **LW-PLA**



Stress Perpendicular to layering direction: 3,380 psi Stress Parallel to layering direction: 6,120 psi

## **Dove Tail Male Construction**

- > Dimensions:
  - ➤ Minimum thickness: 0.5 inches
  - > Maximum thickness: 1 inch
  - ➤ Length: 5.25 inches
- ➤ Characteristics
  - > Follows curvature of the airfoil
  - Edges are rounded



## **Dove Tail Female Construction**

- > Dimensions:
  - ➤ Minimum opening: 0.5001 inches
  - ➤ Maximum thickness: 1.001 inch
  - ➤ Length: 5.25 inches
- Characteristics
  - > Follows curvature of the airfoil
  - > Edges are rounded

