#### FINAL DESIGN REVIEW AIR BEARING UPGRADE FOR SHPB EXPERIMENT

#### Group 1

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#### Sponsored by Eglin Air Force Research Laboratory

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#### December 8, 2011





#### Overview

- Introduction
- Needs Assessment
- Objectives
- Design Concepts
- Decision Matrix
- Final Concept

- Cost Analysis
  - System Analysis
  - Environment & Safety Concern
    - Safety Concerns
  - Remaining Schedule
  - Questions

### Introduction – Overall Design

Design small scale SHPB system



#### Introduction - Shock Wave



#### Needs Assessment

"Warhead design engineers and material scientists" require mechanical property information under high deformation rates of loading on a wide variety of materials that have military significance. The most generally accepted technique for gathering such information is the SHPB experiment. The Air Force Research Laboratory's Damage Mechanisms Branch has operated such an experiment for approximately 30 years. The Damage Mechanisms Branch has a requirement to replace the current bearing system with a new air bearing design."

### **Objectives**

- Analyze SHPB design based on use of air bearings
- Provide analysis of:
  - Hardware cost
  - Interface requirements
  - Installation procedures
  - Impact on bar geometry
- Provide assessment of strain gauge technology
- Develop procedure to align bars
- Design a working prototype to show knowledge of system

# Striker Bar Mechanism Designs

- Pendulum
  - Cost Efficient, < \$50</li>
  - Scalable
  - Easily Built

#### Solenoid

- Consistent Velocities
- Feasible, <\$100</li>
- Reliable Function





# Incident & Transmission Bar Selection

- 8
  - Company: McMaster-Carr
    - 1566 Case Hardened Steel
    - 36" length
    - 0.5"diameter
      - \$20.00
    - 0.75" diameter
      - \$30.00
    - High Tolerances
      - Diameter: 0.0005" to -0.001"
      - Straightness: 0.002" per ft



# **Air Bushings Selection**

- Companies
  - New Way Air Bearings
  - Nelson Air Corp.
- Sizes
  - 0.5" Bushings
    - \$210.00 each (New Way)
    - \$262.00 each (Nelson)
  - 0.75" Bushings
    - \$265.00 each (New Way)
    - \$331.00 each (Nelson)



# Air Supply & Manifold Design

- 10
- Air Supply
  - Compressed Tank
  - Compressor



- Steel Pipe
  - < 100 psi
- Bushing Supply
  - 4 Valves
- System Purge
  - 1 Valve







#### **Base Structure Selection**

- I-Beam
  - Scalable
  - High Strength
  - Cost < \$100
  - Heavy
- T-Slotted Framing
  - Scalable
  - Rigid
  - Cost < \$150
  - Lightweight





# Air Bushing Alignment Method

- Center Bore Alignment
  - Simple insert and check
  - Scalable
  - Accurate
- Exterior Mount Alignment
  - Must be remounted for each bearing
  - Scalable
  - Accurate





# Momentum Trap Designs

- Custom Impact Bumper
  - Simple
  - Durable
  - Cost < \$30
  - Easily Scaled

#### Pre-Manufactured Bumper

- Readily Available
- Various styles
- \$30 range
- Less easily replaced





# **Strain Gage Selection**

- Foil Strain Gages
  - Company : Vishay Micro-Measurements
  - Durable
  - Proven
  - Cost : \$20 per 10 gages
- Semiconductor Gage
  - Company : Micron-Instruments
  - Higher Sensitivity
  - Low Durability
  - Cost: \$80 to \$200 for 8 gages





#### Data Acquisition System Selection

- Software
  - LabVIEW
    - User friendly
    - Quick setup
    - Available at COE
- Hardware
  - Expensive
    - > \$500 for NI platforms
  - Solutions
    - Make use of hardware available at COE or Eglin AFB.
    - Purchase hardware.



#### **Decision Matrix**

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		Lost -	ien, v	simo <sup>r</sup> .	Durab icity	Portab	sility +	Sility	Data Que	lase of a	<b>1</b> 158 -	Score
Base	Weight	<u>0.3</u>	<u>0.2</u>	<u>N/a</u>	<u>0.2</u>	<u>N/a</u>	<u>0.3</u>	N/a	<u>N/a</u>	N/a	N/a	
	I-beam	3	2		4		4					3.3
	T-slot	5	4		4		5					<u>4.6</u>
Bushing	Weight	0.2	<u>0.2</u>	<u>0.1</u>	<u>N/a</u>	<u>0.2</u>	<u>0.1</u>	<u>N/a</u>	<u>0.2</u>	<u>N/a</u>	N/a	
	New Way	3	5	5		5	5		5			4.6
	Nelson	2	5	5		4	5		4			4
Strain Gauges	Weight	0.2	<u>N/a</u>	0.1	N/a	0.2	N/a	N/a	N/a	0.3	0.2	
	Foil	4		4		5				3	4	3.9
	Semiconductor	3		5		4				5	4	4.2
<u>Bar *</u>	Weight	<u>0.1</u>	<u>0.2</u>	<u>0.1</u>	<u>N/a</u>	<u>0.2</u>	<u>0.1</u>	<u>N/a</u>	<u>0.2</u>	<u>0.1</u>	<u>N/a</u>	
	1/2 inch	4	4	5		5	4		3	3		4
	3/4 inch	3	4	5		5	4		4	4		4.2
<u>Striker Bar</u>	Weight	0.2	<u>0.1</u>	<u>N/a</u>	<u>0.2</u>	<u>0.2</u>	<u>N/a</u>	<u>N/a</u>	<u>0.3</u>	<u>N/a</u>	<u>N/a</u>	
	Solenoid	3	4		3	5			5			4.1
	Pendulum	4	4		5	4			3			3.9
<u>Air Manifold</u>	Weight	0.2	<u>0.1</u>	<u>0.2</u>	<u>0.1</u>	<u>0.2</u>	0.2	<u>N/a</u>	<u>N/a</u>	<u>N/a</u>	<u>N/a</u>	
	Horizontal	3	3	3	5	4	3					<mark>3.4</mark>
	Declined	3	3	3	3	4	3					3.2
Bearing Alignmer	n Weight	0.2	<u>N/a</u>	<u>N/a</u>	<u>0.1</u>	<u>N/a</u>	<u>N/a</u>	<u>0.2</u>	<u>0.4</u>	<u>N/a</u>	<u>0.1</u>	1
	Insert	3			2			4	5		5	4.1
	Mounted	3			4			3	4		3	3.5
Momentum Trap	Weight	0.2	<u>0.1</u>	<u>0.1</u>	<u>0.1</u>	0.25	<u>N/a</u>	0.15	<u>N/a</u>	N/a	<u>0.1</u>	
	Custom	4	4	3	4	4		4			4	3.9
	Prefabricated	4	3	3	4	3		3			4	3.4

### **Final Concept**

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Base:	T-Slotted Framing
• Striker Mechanism:	Solenoid
Bars:	0.75" 1566 Steel*
• Bushings:	0.75" New Way
• Bushing Alignment:	Center-Bore Method
• Momentum Trap:	Custom Made
Air Manifold:	Horizontal w/ Purge
Strain Gages:	Foil Type*
• DAQ:	National Instruments

#### Final Concept Model



### **Cost Analysis**

Remaining

\$855.20

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Item	Quantity	Unit Cost	Total Cost
0.75 inch Air Bushings	4	\$331.00	\$1,060.00
Solenoid	1	\$69.94	\$69.94
T-slot Framing 1 1/2 inch (96 inch length)	2	\$48.15	\$96.30
T-slot Framing 1 1/2 inch (4 foot length for 6 inch braces)	1	\$25.15	\$25.15
Incident & Transmission Bar: 1566 Steel Bar 0.75 inch (36inch length)	2	\$29.42	\$58.84
Air Manifold (72 inches)	1	\$16.34	\$16.34
T-slot Framing 1 1/2 inch (24 inch length) For stability	1	\$13.98	\$13.98
Striker Bar: 1566 Steel Bar 0.75 inch (12inch length)	1	\$10.25	\$10.25
Momentum Trap	1	\$30.00	\$30.00
DAQ			\$0.00
Air supply			\$0.00

### System Analysis

- Plastic deformation of copper specimen is most important factor in analysis
- Yield Stress
   Minimum Force
   Striker Bar Momentum (mass)
  - Duration of Impact
- Minimum Striker Bar Velocity

### System Analysis

Secondary Analysis Requirements:

#### Low Cost

- Many solenoid options require the use of <u>optimization</u> <u>methods</u>
- Elastic Deformation in Steel (Maximum Striker bar Velocity)
  - Stress concentrations require use of Finite Element Modeling

#### System Analysis: Solenoid Optimization



#### System Analysis: Stress Multiplication



#### System Analysis: Maximum Stress



# **Environment & Safety Concerns**

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- No Environmental Hazards
- Safety Concerns
  - Pinching or crushing of fingers between bars.
  - Safety Shields



### **Remaining Schedule**



# 27 Questions? Comments?

Our Thanks to: Dr. House, Eglin AFRL Dr. Hovsapian, FSU COE Dr. Kosaraju, FSU COE

□ Stress  $\sigma = F/A$ 

- $\Box Strain \qquad \epsilon = (Li Lo) / Lo$
- □ Gauge Factor  $GF = [(Ri Ro) / Ro] / \epsilon$
- □ Data Strain  $\epsilon$  (Ri) = [(Ri Ro) / Ro] / GF

Strain in Specimen:

$$d\epsilon_{avg} / dt = (c_b / L_s) * (\epsilon_{I-} - \epsilon_R - \epsilon_T)$$

□ Integration:

$$\varepsilon_{s} = (C_{b} / L_{s}) * \int_{0}^{t} [(\varepsilon_{I} - \varepsilon_{R} - \varepsilon_{T}) * dt]$$

Strain through the specimen

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Strain energy for each wave

Kinetic energy = 
$$0.5 * m * v^2$$

Initial	$E_{I} = 0.5^{*} A_{B}^{*} C_{B}^{*} E_{B}^{*} T^{*} \epsilon_{I}^{2}$
Reflected	$E_r = 0.5^* A_B^* C_B^* E_B^* T^* \epsilon_R^2$
Transmitted	$E_{t} = 0.5^{*} A_{B}^{*} C_{B}^{*} E_{B}^{*} T^{*} \epsilon_{T}^{2}$

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Strain energy

#### $\delta S_{E} = E_{I} - E_{R} - E_{T}$

Plastic Energy absorbed by specimen

$$E_s = 2 * \delta S_E$$

### **Velocity Calculations**

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The velocity of the striker bar is needed The only requirement is that the specimen plasticaly deform while the incident and transmitter bars are only loaded elasticaly The following equations show the process

$$\sigma_{yc} := 70 \text{MPa}$$

$$\text{Area}_{c} := \pi \cdot \left(\frac{.4}{2}\text{in}\right)^{2} = \mathbf{I} \cdot \text{in}^{2}$$

$$\text{F} := \sigma_{yc} \cdot \text{Area}_{c} = \mathbf{I} \cdot \text{kN}$$

Yield stress of copper

Area of the copper

Force Required to reach Yield

Next the mass of the steel bar is computed

$$\rho := 7.85 \frac{\text{gm}}{\text{cm}^3}$$
$$v := \pi \cdot \left(\frac{0.75}{2}\right)^2 \text{ in}^2 \cdot 6\text{ in} = \mathbf{1} \cdot \text{ in}^3$$
$$\text{mass} := \mathbf{v} \cdot \rho = \mathbf{1}$$

Density of steel

Volume of the 3/4 inch diameter, 6 in striker bar

Mass of the striker bar

#### **Velocity Calculations**

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Next the amount of time the striker bar will impact the incident bar

$c := 6100 \frac{m}{s}$ $L := 6in$	Speed of wave propogation in steel Length of Striker bar
$\mathbf{t} := 2 \cdot \frac{\mathbf{L}}{\mathbf{c}} = \mathbf{I} \cdot \mathbf{s}$	Pressure wave propogating down the strikerbar and returning = 2 x length/speed
t = ∎·μs	Duration of impact

Finaly the minimum velocity of the striker bar needed to plasticaly deform the specimen

$$\mathbf{V} := \frac{\mathbf{F}}{\max} \cdot \mathbf{t} = \mathbf{I}$$



Minimum velocity of striker bar needed to plasticaly deform the copper specimen

#### **Velocity Calculations**

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Acc :=  $\frac{310\text{ozf}}{\text{mass}} = \mathbf{I}$ 

 $L_{sol} := 1in$ 

$$D = D_0 + V_0 \cdot t + .5A_{cc} \cdot t^2$$

$$\operatorname{time}_{\operatorname{sol}} := \left(\frac{L_{\operatorname{sol}}}{0.5 \operatorname{Acc}}\right)^{.5} = \bullet$$

$$V_{stkr} := Acc \cdot time_{sol} = \mathbf{I} \cdot \frac{mi}{hr}$$

Acceleration available from a chosen solenoid

Length of piston with given force

Generic dynamic position equation

Derived time, from previous equation

Calculated velocity from given solenoic

Force<sub>striker.sol</sub> := 
$$\frac{V_{stkr} \cdot mass}{t} = \mathbf{I} \cdot kN$$

Maximum force transfered from solence

#### Weak Formulation for FEA

$$\rho \cdot \mathbf{A} \cdot \frac{\mathbf{d}^2}{\mathbf{dt}^2} \mathbf{T} - \frac{\mathbf{d}}{\mathbf{dx}} \left[ \mathbf{E} \cdot \mathbf{A} \cdot \left( \frac{\mathbf{d}}{\mathbf{dx}} \mathbf{u} \right) \right] - \mathbf{f}(\mathbf{x}, \mathbf{t}) = 0$$

$$\int \left[ w \cdot \rho \cdot A \cdot \frac{d^2}{dt^2} T - w \cdot \frac{d}{dx} \left[ E \cdot A \cdot \left( \frac{d}{dx} u \right) \right] - w f(x, t) \right] d(x, t) = 0$$

$$\int \left[ -(\rho \cdot A) \cdot \left( \frac{d}{dt} w \right) \cdot \left( \frac{d}{dt} T \right) - w f(x,t) \right] d(x,t) + w \cdot A \cdot \rho \cdot \left( \frac{d}{dt} T \right) = 0$$

#### Weak Formulation for FEA

 $[\mathsf{K}]^*\{\mathsf{u}\} + [\mathsf{C}]^*\{\tilde{\mathsf{u}}\} + [\mathsf{M}]^*\{\tilde{\upsilon}\} = \{F\}$ 



$$M_{ij} = \int_{xa}^{xb} c_0(x) \cdot \psi_i \cdot \psi_j \, dx$$