Test Results N=8 teeth **Working Fluid: AIR**

Nomenclature:

Cc = Empirical coefficient

Cr = Empirical coefficient

Ct = Empirical coefficient

Af = Flow Area

T = Temperature (high P side)

t = Tooth thickness

N = number of teeth

P = Pressure

Pr = Pressure ratio

p = Tooth spacing

R = Gas Constant

Re = Reynolds Number

r = radius

 δ = gap diameter

 ρ = density (high P side)

Governing Equations:

$$\mathbf{m}_{\text{dot}} = \pi \cdot 2 \cdot \mathbf{r}_{0} \cdot \delta \cdot \mathbf{C}_{t} \cdot \mathbf{C}_{c} \cdot \mathbf{C}_{r} \cdot \rho_{0} \cdot \sqrt{\mathbf{R} \cdot \mathbf{T}}$$

$$m_{dot} = \rho AV$$

$$C_t = 2.143 \cdot \frac{\ln(N) - 1.464}{n - 4.322} \cdot (1 - P_r)^{0.375 \cdot P_r}$$
 $R = C_p - C_v$

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$$C_{r} = 1 - \frac{1}{3 + \left(\frac{54.3}{1 + 100 \cdot \frac{\delta}{t}}\right)^{3.45}}$$

$$C_c = 1 + X_1 \cdot \frac{\frac{\delta}{p} - X_2 \cdot \ln\left(1 + \frac{\delta}{p}\right)}{1 - X_2}$$

$$A = \pi \left(r_0^2 - r_i^2 \right)$$

$$Re = \frac{\rho VD}{\mu} = \frac{VD}{v}$$

$$X_1 = 15.1 - 0.05255 \cdot e^{0.507 \cdot (12 - N)}$$

$$X_2 = 1.058 + .0218 \cdot N$$
 $N < or = 12$

$$X_1 = 13.15 + .1625N$$

$$X_2 = 1.32$$

Seal Dimensions: Impeller Labyrinth Gap size .18mm

$$r_0 := \frac{68}{2}$$
mm $p := 1$ mm $\delta := 0.18$ mm $t := .5$ mm $N_{teeth} := 8$

$$\delta = 0.18 \text{mm}$$

$$t := .5 \text{mm}$$

$$N_{teeth} := 8$$

ro is the inner diameter of the seal but the outer diameter of the teeth

$$r_i := r_o - \delta$$

$$A_{flow} := \pi \cdot (r_o^2 - r_i^2)$$
 $A_{flow} = 38.351 \text{ mm}^2$

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Air Properties at Testing Conditions

kJ := 1000J

$$T_{hp} := (21 + 273.15)K$$
 $P_{L} := .1048MPa$ $P_{H} := .1772MPa$ $P_{r} := \frac{P_{L}}{P_{H}}$

$$P_{L} := .1048MP_{2}$$

$$P_H := .1772MPa$$

$$P_r := \frac{P_L}{P_H}$$

$$\rho_0 := 2.1484 \frac{\text{kg}}{\text{m}^3}$$

$$C_{p} := 1.0076 \frac{kJ}{kg \cdot k}$$

$$C_{V} := .71781 \frac{kJ}{kg \cdot K}$$

$$\rho_{0} := 2.1484 \frac{\text{kg}}{\text{m}^{3}} \qquad \qquad C_{p} := 1.0076 \frac{\text{kJ}}{\text{kg} \cdot \text{K}} \qquad C_{v} := .71781 \frac{\text{kJ}}{\text{kg} \cdot \text{K}} \qquad v := .083540 \frac{\text{cm}^{2}}{\text{s}}$$

$$\mathop{\mathbb{R}} := \mathrm{C}_p - \mathrm{C}_v$$

Mass Flow Calculation:

$$C_{r} := 1 - \frac{1}{3 + \left(\frac{54.3}{1 + 100 \cdot \frac{\delta}{t}}\right)^{3.45}}$$

$$C_{r} = 0.852$$

$$C_{t} := 2.143 \cdot \frac{\ln(N_{teeth}) - 1.464}{N_{teeth} - 4.322} \cdot (1 - P_{r})^{0.375 \cdot P_{r}}$$

$$C_{t} = 0.294$$

$$\mathbf{X}_1 := 15.1 - 0.05255 \cdot \mathrm{e}^{0.507 \cdot \left(12 - \mathrm{N}_{\mathrm{teeth}}\right)}$$

$$\mathbf{X}_1 = 14.701$$

$$X_2 := (1.058 + .0218 \cdot N_{teeth})$$
 $X_2 = 1.232$

$$C_c := 1 + X_1 \cdot \frac{\frac{\delta}{p} - X_2 \cdot \ln\left(1 + \frac{\delta}{p}\right)}{1 - X_2}$$

$$C_c = 2.517$$

$$\mathbf{m}_{\text{dot}} := \pi \cdot 2 \cdot \mathbf{r}_{0} \cdot \delta \cdot \mathbf{C}_{t} \cdot \mathbf{C}_{c} \cdot \mathbf{C}_{r} \cdot \rho_{0} \cdot \sqrt{\mathbf{R} \cdot \mathbf{T}_{hp}}$$

$$\mathbf{m}_{\text{dot}} = 0.015 \frac{\mathbf{kg}}{\mathbf{s}}$$